

Arithmetic mean

$$\bar{x} = \frac{1}{N} \sum x_i$$

Standard deviation

$$s = \sqrt{\frac{\sum (x_i - \bar{x})^2}{N-1}}$$

Torsion

$$\theta = \frac{T.L}{G.J}$$

Gaussian (normal) distribution

$$f(x) = \frac{1}{s\sqrt{2\pi}} \exp\left[-\frac{1}{2}\left(\frac{x-\bar{x}}{s}\right)^2\right]$$

Closed thin-walled tubes

$$\tau = \frac{T}{2.A.t}$$

A is the area enclosed by the section median line

$$\theta_1 = \frac{T.L}{4.G.A^2.t}$$

L is the perimeter of the section median line

twist per unit length

Lognormal distribution

$$f(x) = \frac{1}{x.\bar{s}\sqrt{2\pi}} \exp\left[-\frac{1}{2}\left(\frac{\ln x - \bar{x}}{\bar{s}}\right)^2\right]$$

$$\bar{s} = \sqrt{\ln(1+C^2)} \quad C = \frac{s}{\bar{x}}$$

Open thin-walled sections

$$\tau = G.\theta_1.C = \frac{3.T}{L.C^2}$$

Weibull distribution

$$R(x) = \exp\left[-\left(\frac{x-x_0}{\theta-x_0}\right)^b\right]$$

Stress concentration

$$K_t = \frac{\sigma_{max}}{\sigma_0} \quad K_{ts} = \frac{\tau_{max}}{\tau_0}$$

Stress transformation

$$\sigma = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\phi + \tau_{xy} \cdot \sin 2\phi$$

Stresses in cylinders

$$\sigma_t = \frac{P_i r_i^2 - P_o r_o^2 - \frac{r_i^2 r_o^2}{r^2} (P_o - P_i)}{r^2}$$

$$\sigma_r = \frac{P_i r_i^2 - P_o r_o^2 + \frac{r_i^2 r_o^2}{r^2} (P_o - P_i)}{r_o^2 - r_i^2}$$

$$\sigma_c = \frac{P_i r_i^2}{r_o^2 - r_i^2}$$

$$\tau = -\frac{\sigma_x - \sigma_y}{2} \sin 2\phi + \tau_{xy} \cdot \cos 2\phi$$

Octahedral stresses

$$\tau_{oct} = \frac{1}{3} \left[ (\sigma_x - \sigma_y)^2 + (\sigma_y - \sigma_z)^2 + (\sigma_z - \sigma_x)^2 + 6(\tau_{xy}^2 + \tau_{xz}^2 + \tau_{yz}^2) \right]^{1/2}$$

$$\sigma_{oct} = \frac{1}{3} (\sigma_x + \sigma_y + \sigma_z)$$

Rotating rings

$$\sigma_t = f.\omega^2 \left(\frac{3+\nu}{8}\right) \left(r_i^2 + r_o^2 + \frac{r_i^2 r_o^2}{r^2} - \frac{1+3\nu}{3+\nu} r^2\right)$$

Shear stresses in beams

$$\tau = \frac{V.Q}{I.t}$$

the first moment of the area

$$\sigma_r = f.\omega^2 \left(\frac{3+\nu}{8}\right) \left(r_i^2 + r_o^2 - \frac{r_i^2 r_o^2}{r^2} - r^2\right)$$

Press and shrink fits

$$\sigma_{it} = -p \frac{R^2 + r_i^2}{R^2 - r_i^2} \quad \sigma_{ot} = p \frac{r_o^2 + R^2}{r_o^2 - R^2}$$

Torque

$$T = \frac{H}{\omega} = 9550 \frac{H}{n}$$

N.m      kW      rpm

$$p = \frac{E.s}{R} \left[ \frac{(r_o^2 - R^2)(R^2 - r_i^2)}{2.R^2(r_o^2 - r_i^2)} \right]$$

# Temperature effects

$$\epsilon_x = \epsilon_y = \epsilon_z = \alpha (\Delta T) \quad \alpha \text{ is the coefficient of thermal expansion}$$

## Curved members in flexure

$$\sigma = \frac{M \cdot y}{A \cdot e \cdot (r_n - y)} \quad r_n = \frac{A}{\int \frac{dA}{r}} \quad e = R - r_n$$

## Contact stresses

### 1) Spherical contact

$$\sigma_1 = \sigma_2 = \sigma_x = \sigma_y = -P_{max} \left[ \left( 1 - \left| \frac{z}{a} \right| \tan^{-1} \frac{1}{|z/a|} \right) (1 + \nu) - \frac{1}{2 \left( 1 + \frac{z^2}{a^2} \right)} \right]$$

$$\sigma_3 = \sigma_z = \frac{-P_{max}}{1 + \frac{z^2}{a^2}} \quad P_{max} = \frac{3F}{2\pi a^2} \quad a = \sqrt[3]{\frac{3F(1-\nu_1^2)/E_1 + (1-\nu_2^2)/E_2}{8 \left( \frac{1}{d_1} + \frac{1}{d_2} \right)}}$$

### 2) Cylindrical contact

$$\sigma_x = -2\nu P_{max} \left( \sqrt{1 + \frac{z^2}{b^2}} - \left| \frac{z}{b} \right| \right) \quad P_{max} = \frac{2F}{\pi \cdot b \cdot L}$$

$$\sigma_y = -P_{max} \left( \frac{1 + 2 \frac{z^2}{b^2}}{\sqrt{1 + \frac{z^2}{b^2}}} - 2 \left| \frac{z}{b} \right| \right) \quad b = \sqrt{\frac{2F(1-\nu_1^2)/E_1 + (1-\nu_2^2)/E_2}{\pi \cdot L \left( \frac{1}{d_1} + \frac{1}{d_2} \right)}}$$

$$\sigma_3 = \sigma_z = \frac{-P_{max}}{\sqrt{1 + \frac{z^2}{b^2}}}$$

## Deflection due to bending

$$\frac{M}{EI} = \frac{d^2 y}{dx^2}$$

## Strain energy

$$U = \frac{F^2 \cdot L}{2AE} + \frac{T^2 \cdot L}{2GJ} + \frac{F^2 \cdot L}{2AG} + \int \frac{M^2 dx}{2EI} \quad I = k^2 \cdot A$$

tension compression      torsion      direct shear      bending

$$\left( \frac{L}{k} \right)_1 = \left( \frac{2\pi^2 CE}{S_y} \right)^{1/2}$$

bending shear  $\Rightarrow U_s = \int \frac{C \cdot v^2 dx}{2 \cdot A \cdot G}$

2) Johnson  $\left( \frac{L}{k} \right) \ll \left( \frac{L}{k} \right)_1$

$$\frac{P_{cr}}{A} = S_y - \left( \frac{S_y \cdot L}{2\pi k} \right)^2 \cdot \frac{1}{CE}$$

## Castigliano's theorem

$$\delta_i = \frac{\partial U}{\partial F_i}$$

### 3) Eccentric loading (secant)

$$\frac{P}{A} = \frac{S_{yc}}{1 + (e \cdot c / k^2) \sec \left[ (L/2k) \sqrt{P/AE} \right]}$$

## Deflection of curved members

bending  $\Rightarrow U_b = \int \frac{M^2 d\theta}{2 \cdot A \cdot e \cdot E}$

## Buckling

### 1) Euler

$$P_{cr} = \frac{C \cdot \pi^2 EI}{L^2}$$

4) Struts  $\Leftrightarrow \left(\frac{L}{k}\right) \leq \left(\frac{L}{k}\right)_2 \quad \left(\frac{L}{k}\right)_2 = 0.282 \left(\frac{A \cdot E}{P_{cr}}\right)^{1/2}$

$$\sigma_c = \frac{P}{A} \left(1 + \frac{e \cdot y}{k^2}\right)$$

Deflection of energy dissipative assemblies

1) Single orifice

$$F = c \cdot \dot{x}_0 \exp\left(-\frac{c g t}{W}\right) = c \cdot \dot{x}_0 - \frac{c^2 g x}{W} \quad x_s = \frac{W \cdot \dot{x}_0}{c \cdot g}$$

$$E_k = \frac{1}{2} \frac{W \cdot \dot{x}_0^2}{g} \quad I = \frac{W \cdot \dot{x}_0}{g}$$

2) Multiple orifices

$$x_s = \frac{\dot{x}_0^2}{2 \cdot \ddot{x}_0} \quad E_k = \frac{W \cdot \dot{x}_0^2}{2 \cdot g} \quad I = \frac{W}{g} \cdot \dot{x}_0 \quad F = \frac{W \cdot \ddot{x}_0}{g}$$

Suddenly-applied loading

$$y = \frac{v}{(k g / W)^{1/2}} \sin\left(\left(\frac{k g}{W}\right)^{1/2} \cdot t\right) \quad \leftarrow \quad \begin{array}{c} \text{|||||} \\ \text{EI, L} \\ \text{---} \rightarrow y \\ \text{W} \\ \text{v} \end{array}$$

$$y = \left[\left(\frac{W}{k}\right)^2 + \frac{2Wh}{k}\right]^{1/2} \cos\left[\left(\frac{k g}{W}\right) t' - \phi\right] + h + \frac{W}{k} \quad \leftarrow \quad \begin{array}{c} \text{W} \\ \downarrow \\ \text{---} \text{---} \text{---} \text{---} \\ \text{h} \\ \text{---} \text{---} \text{---} \text{---} \\ \text{|||||} \\ \text{k} \end{array}$$

$$\phi = \frac{\pi}{2} + \tan^{-1}\left(\frac{W}{2kh}\right)^{1/2}$$

$$t' = t - t_1 \quad t_1 = \left(\frac{2h}{g}\right)^{1/2}$$

$$F = k \cdot \delta = W + W \left[1 + \left(\frac{2hk}{W}\right)\right]^{1/2}$$

Notch sensitivity

$$q = \frac{K_f - 1}{K_t - 1}$$

Paris equation (Life prediction)

$$\frac{da}{dN} = A (\Delta K_I)^n$$

Maximum shear stress (Tresca) hypothesis

$$\tau_{max} \geq \frac{S_y}{2} \quad \text{OR} \quad \sigma_1 - \sigma_3 \geq S_y \Rightarrow \text{failure occurs}$$

von Mises hypothesis

$$\sigma' = \left[ \frac{(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2}{2} \right]^{1/2} = \frac{1}{\sqrt{2}} \left[ (\sigma_x - \sigma_y)^2 + (\sigma_y - \sigma_z)^2 + (\sigma_z - \sigma_x)^2 + 6(\tau_{xy}^2 + \tau_{yz}^2 + \tau_{zx}^2) \right]^{1/2} \geq S_y \quad \text{failure occurs}$$

Internal-friction hypothesis

$$\sigma_1 - \sigma_3 = S_y \quad \frac{\sigma_1}{S_t} - \frac{\sigma_3}{S_c} = 1 \quad \sigma_1 \geq 0 \quad \sigma_3 \leq 0 \Rightarrow \text{failure occurs}$$

$$S_{sy} = \frac{S_{yt} S_{yc}}{S_{yt} + S_{yc}}$$

larger first-cycle-yielding locus

Maximum normal stress hypothesis

$$n \cdot \sigma_1 = S_t \quad n \cdot \sigma_3 = -S_c$$

$$\frac{S_a}{S_{yt}} + \frac{S_m}{S_{yt}} = 1$$

Mohr hypothesis

$$\sigma_A = \frac{S_{ut}}{n} \quad 0 < \sigma_A < S_{ut} \quad 0 < \sigma_B < S_{ut}$$

Modified Goodman line

$$\frac{\sigma_a}{S_e} + \frac{\sigma_m}{S_{ut}} = \frac{1}{n}$$

$$\frac{\sigma_A}{S_{ut}} - \frac{\sigma_B}{S_{uc}} = \frac{1}{n} \quad 0 < \sigma_A < S_{ut} \quad -S_{uc} < \sigma_B < 0$$

Strain-life relationships

$$\frac{\Delta \epsilon}{2} = \underbrace{\frac{\sigma_F'}{E} (2N)^b}_{\text{high cycle fatigue}} + \underbrace{\epsilon_F' (2N)^c}_{\text{low cycle fatigue}}$$

Gerber locus

$$\frac{n \sigma_a}{S_e} + \left( \frac{n \sigma_m}{S_{ut}} \right)^2 = 1$$

DE-elliptic locus

$$\left( \frac{n \sigma_a}{S_e} \right)^2 + \left( \frac{n \sigma_m}{S_y} \right)^2 = 1$$

Endurance limit

$$S_e = k_a k_b k_c k_d k_e S_e'$$

Module

$$m = d/N$$

d is pitch diameter

Fluctuating stresses

$$\sigma_m = \frac{\sigma_{max} + \sigma_{min}}{2}$$

$$p = \frac{\pi \cdot d}{N}$$

p is circular pitch

$$\sigma_a = \left| \frac{\sigma_{max} - \sigma_{min}}{2} \right|$$

Helical gears

$$p_n = p_t \cdot \cos \psi$$

$$p_x = \frac{p_t}{\tan \psi}$$

$$\cos \psi = \frac{\tan \phi_n}{\tan \phi_t}$$

Soderberg line

$$\frac{\sigma_a}{S_e} + \frac{\sigma_m}{S_y} = \frac{1}{n}$$

ψ: helix angle φ: pressure angle

$$N' = \frac{N}{\cos^3 \psi}$$

virtual number of teeth

Straight bevel gears

$$\tan \gamma = \frac{N_P}{N_G} \quad \tan \Gamma = \frac{N_G}{N_P}$$

Worm gears

$$d_g = \frac{N_g \cdot P_t}{\pi}$$

$$\frac{C^{0.875}}{3} \leq d_w \leq \frac{C^{0.875}}{1.7}$$

$$\tan \lambda = \frac{L}{\pi d_w}$$

Gear trains

$$n_L = e \cdot n_F$$

$$e = \frac{\text{product of driving tooth numbers}}{\text{product of driven tooth numbers}}$$

Planetary gear trains

$$e = \frac{n_L - n_A}{n_F - n_A}$$

Force analysis - worm gearing

$$W^x = W \cdot \cos \phi_n \cdot \sin \lambda$$

$$W^y = W \cdot \sin \phi_n$$

$$W^z = W \cdot \cos \phi_n \cdot \cos \lambda$$

$$W_{wt} = -W_{ga} = W^x$$

$$W_{wr} = -W_{gr} = W^y$$

$$W_{wa} = -W_{gt} = W^z$$

$$\eta = \frac{\cos \phi_n - f \cdot \tan \lambda}{\cos \phi_n + f \cdot \cot \lambda}$$

efficiency (worm gearing)

Choosing module

1) Pitting

$$m_n = 0.9 \sqrt[3]{\frac{K_A \cdot K_V \cdot T_1 \cdot E (i+1) \cos^4 \psi}{N_1^2 \cdot P_{em}^2 \cdot i \cdot \epsilon_\alpha \cdot F_w}}$$

2) Fracture

$$m_n = 0.6 \sqrt[3]{\frac{K_A \cdot K_V \cdot T_1 \cdot \delta \cdot \cos \psi}{N_1 \cdot \sigma_{em} \cdot \epsilon_\alpha \cdot F_w}}$$

Force analysis - helical gearing

$$W_r = W \cdot \sin \phi_n$$

$$W_t = W \cdot \cos \phi_n \cdot \cos \psi$$

$$W_a = W \cdot \cos \phi_n \cdot \sin \psi$$

$$T = \frac{d}{2} \cdot W_t$$

Force analysis - bevel gearing

$$W_t = \frac{T}{r_{av}} \quad W_r = W_t \cdot \tan \phi \cos \delta$$

$$W_a = W_t \cdot \tan \phi \sin \delta$$

Power screws

$$T = \frac{F \cdot d_m}{2} \left( \frac{L + \pi \mu d_m}{\pi d_m - \mu L} \right) + \frac{F \cdot \mu_c d_c}{2} \quad \leftarrow \text{raising the load}$$

$$T = \frac{F \cdot d_m}{2} \left( \frac{\pi \mu d_m - L}{\pi d_m + \mu L} \right) + \frac{F \cdot \mu_c \cdot d_c}{2} \quad \leftarrow \text{lowering the load}$$

Tension connections

$$k_b = \frac{A_d A_t E}{A_d \cdot L_t + A_t \cdot L_d}$$

$$k_m = \frac{\pi \cdot E \cdot d \cdot \tan \alpha}{2 \cdot \ln \frac{(L \cdot \tan \alpha + d_w - d)(d_w + d)}{(L \cdot \tan \alpha + d_w + d)(d_w - d)}}$$

$$T = (0.20) \cdot F_i \cdot d$$

$$F_i = \begin{cases} 0.75 F_p & \text{for reused connections} \\ 0.90 F_p & \text{for permanent connections} \end{cases}$$

Bearing life

$$F_e \cdot L^{1/a} = C \quad \begin{matrix} a=3 & \text{for ball bearings} \\ a=10/3 & \text{" roller "} \end{matrix}$$

$$C_{10} \cdot L_{10}^{1/a} = F \cdot L^{1/a}$$

$$C_{10} = F_D \cdot \left[ \frac{x_D}{x_0 + (\theta - x_0) (\ln 1/R_D)^{1/b}} \right]^{1/a}$$

$$R = \exp \left[ - \frac{\left[ x_D - x_0 \left( \frac{C'_{10}}{f_A \cdot F_D} \right)^a \right]^b}{(\theta - x_0) \left( \frac{C'_{10}}{f_A \cdot F_D} \right)^a} \right]$$

C'10: basic load rating

Journal bearings

$$S = \left( \frac{r}{c} \right)^2 \left( \frac{\mu \cdot N}{P} \right)$$

Welded, brazed, bonded joints

$$I = (0.707) h \cdot I_u$$

Mechanical springs

$$\tau = \frac{8 \cdot F \cdot D}{\pi \cdot d^3} + \frac{4F}{\pi d^2} \quad C = \frac{D}{d} \quad K_s = \frac{2C+1}{2C} \quad K_B = \frac{4C+2}{4C-3}$$

$$L_0 < \frac{\pi \cdot D}{\alpha} \left[ \frac{2[E-G]}{2G+E} \right]^{1/2} \quad \left| \begin{array}{l} \text{Clutches, brakes} \\ dN = p \cdot b \cdot r \cdot d\theta \\ p = \frac{Pa}{\sin \theta_a} \cdot \sin \theta \end{array} \right.$$

$$w = \pi \sqrt{\frac{k \cdot g}{W}}$$

Internal expanding rim clutches and brakes

$$M_f = \int_{\theta_1}^{\theta_2} f dN (r - a \cdot \cos \theta) \quad M_N = \int_{\theta_1}^{\theta_2} dN (a \cdot \sin \theta)$$

for clockwise  $F = \frac{M_N - M_f}{c} \quad T = \int_{\theta_1}^{\theta_2} f \cdot r \cdot dN$

$$R_x = \int_{\theta_1}^{\theta_2} dN \cdot \cos \theta - \int_{\theta_1}^{\theta_2} f dN \sin \theta - F_x$$

$$R_y = \int_{\theta_1}^{\theta_2} dN \cdot \sin \theta + \int_{\theta_1}^{\theta_2} f dN \cos \theta - F_y$$

for counterclockwise  $F = \frac{M_N + M_f}{c}$

$$R_x = \int_{\theta_1}^{\theta_2} dN \cdot \cos \theta + \int_{\theta_1}^{\theta_2} f dN \sin \theta - F_x$$

$$R_y = \int_{\theta_1}^{\theta_2} dN \sin \theta - \int_{\theta_1}^{\theta_2} f dN \cos \theta - F_y$$

External contracting rim clutches and brakes

$$M_f = \frac{f \cdot Pa \cdot b \cdot r}{\sin \theta_a} \int_{\theta_1}^{\theta_2} \sin \theta (r - a \cdot \cos \theta) d\theta$$

$$M_N = \frac{Pa \cdot b \cdot r \cdot a}{\sin \theta_a} \int_{\theta_1}^{\theta_2} \sin^2 \theta d\theta$$

for clockwise  $F = \frac{M_N + M_f}{c}$

$$R_x = \int dN \cdot \cos \theta + \int f dN \sin \theta - F_x \quad R_y = \int f dN \cos \theta - \int dN \sin \theta + F_y$$

for counterclockwise

$$F = \frac{M_N - M_f}{c} \quad R_x = \int dN \cos \theta - \int f dN \sin \theta - F_x$$

$$R_y = - \int f dN \cos \theta - \int dN \sin \theta + F_y$$

Band type clutches and brakes

$$\frac{P_1}{P_2} = e^{f \cdot \phi} \quad T = (P_1 - P_2) \cdot \frac{D}{2}$$

$$dN = p \cdot b \cdot r \cdot d\theta = P \cdot d\theta \Rightarrow p = \frac{P}{b \cdot r}$$

Axial clutches and brakes

$$F = \frac{\pi \cdot P_a \cdot d}{2} (D - d) \quad \left. \begin{array}{l} F = \frac{\pi \cdot P_a \cdot d}{2} (D - d) \\ T = \frac{\pi \cdot P_a \cdot d}{8} (D^2 - d^2) \end{array} \right\} \text{uniform wear}$$

$$T = \frac{\pi \cdot P_a \cdot d}{8} (D^2 - d^2)$$

$$F = \frac{\pi \cdot P_a}{4} (D^2 - d^2) \quad \left. \begin{array}{l} F = \frac{\pi \cdot P_a}{4} (D^2 - d^2) \\ T = \frac{2 \cdot \pi \cdot f \cdot P}{24} (D^3 - d^3) \end{array} \right\} \text{uniform pressure}$$

$$T = \frac{2 \cdot \pi \cdot f \cdot P}{24} (D^3 - d^3)$$

$$P = P_a$$

Cone clutches and brakes

$$F = \frac{\pi \cdot P_a \cdot d}{2} (D - d) \quad \left. \begin{array}{l} F = \frac{\pi \cdot P_a \cdot d}{2} (D - d) \\ T = \frac{F \cdot f}{4 \cdot \sin \alpha} (D + d) \end{array} \right\} \text{uniform wear}$$

$$T = \frac{F \cdot f}{4 \cdot \sin \alpha} (D + d)$$

$$F = \frac{\pi \cdot P_a}{4} (D^2 - d^2)$$

$$T = \frac{F \cdot f}{3 \cdot \sin \alpha} \frac{D^3 - d^3}{D^2 - d^2}$$

Flexible mechanical elements

$$\frac{F_1 - F_c}{F_2 - F_c} = \exp(f \cdot \phi)$$

$$F_2 - F_c$$

$$F_c = m \cdot r^2 \cdot \omega^2$$

$$F_1 = F_i + F_c + T/D$$

$$F_2 = F_i + F_c - T/D$$