

A shaft is a rotating member used to transmit power or motion.

An axle is a nonrotating member which carries no torque and is used to support rotating wheels, pulleys, etc.

Design considerations \Rightarrow Deflection and rigidity (bending deflection, torsional deflection, slope at bearings and shaft-supported elements, shear deflection due to transverse loading of short shafts); Stress and strength (static strength, fatigue strength, reliability)

Torque-transfer elements \Rightarrow keys, splines, setscrews, pins, press or shrink fits, tapered fits.

Locational-devices (for accurate axial location of the device) \Rightarrow cotter and washer, nut and washer, sleeve, shaft shoulder, ring and groove, setscrew, split hub or tapered two-piece hub, collar and screw, pins.

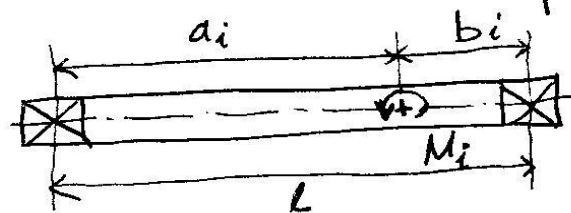
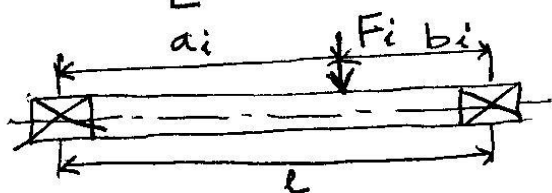
Geometric Constraints

for the left-bearing slope constraint active

$$d = \left| \frac{32n}{3\pi E L \Sigma \theta} \left\{ \left[\sum F_i b_i (b_i^2 - l^2) + \sum M_i (3a_i^2 - 6a_i l + 2l^2) \right]_H^2 + \left[\sum F_i b_i (b_i^2 - l^2) + \sum M_i (3a_i^2 - 6a_i l + 2l^2) \right]_V^2 \right\}^{1/2} \right|^{1/4}$$

for the right-bearing constraint active

$$d = \left| \frac{32n}{3\pi E L \Sigma \theta} \left\{ \left[\sum F_i a_i (l^3 - a_i^3) + \sum M_i (3a_i^2 - l^2) \right]_H^2 + \left[\sum F_i a_i (l^3 - a_i^3) + \sum M_i (3a_i^2 - l^2) \right]_V^2 \right\}^{1/2} \right|^{1/4}$$



$\Sigma \theta \Rightarrow$ absolute value of the allowable slope at the bearing

Strength Constraints

$$\sigma_a = \left| \frac{32 \cdot M_a}{\pi d^3} \right| \quad \tau_{xym} = \frac{16 \cdot T_m}{\pi \cdot d^3}$$

$$\sigma_a' = \sigma_a \quad \tau_m' = \sqrt{3} \tau_{xym}$$

Fatigue - Strength Loci for Shafts

Locus Name

Basic Formula

Soderberg

$$\frac{n \sigma_a}{S_e} + \frac{n \tau_m}{S_y} = 1$$

Goodman

$$\frac{n \sigma_a}{S_e} + \frac{n \tau_m}{S_{ut}} = 1$$

Gerber

$$\frac{n \sigma_a}{S_e} + \left(\frac{n \tau_m}{S_{ut}} \right)^2 = 1$$

ASME elliptic

$$\left(\frac{n \sigma_a}{S_e} \right)^2 + \left(\frac{n \tau_m}{S_y} \right)^2 = 1$$

Bagci

$$\frac{n \sigma_a}{S_e} + \left(\frac{n \tau_m}{S_y} \right)^4 = 1$$

Yielding (Langer) $\frac{n}{S_y} (\sigma_a + \tau_m) = 1$

Shaft Diameter Equation for the DE - Gerber Criterion

$$\sigma_a' = (\sigma_{xa}^2 + 3 \tau_{xya}^2)^{1/2} = \frac{16}{\pi d^3} \sqrt{4 (K_f M_a)^2 + 3 (K_{fs} T_a)^2}$$

$$\tau_m' = (\sigma_{xm}^2 + 3 \tau_{xym}^2)^{1/2} = \frac{16}{\pi d^3} \sqrt{4 (K_f M_m)^2 + 3 (K_{fs} T_m)^2}$$

Critical Speeds

$$\omega_i = \left(\frac{\pi}{L} \right)^2 \sqrt{\frac{EI'}{m}} = \left(\frac{\pi}{L} \right)^2 \sqrt{\frac{\beta EI}{A \cdot \gamma}}$$

for an ensemble of attachments

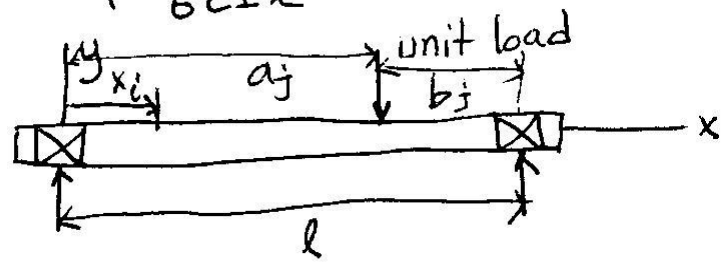
$$\omega_i = \sqrt{\frac{g \cdot \sum w_i \cdot y_i}{\sum w_i \cdot y_i^2}}$$

$w_i \Rightarrow$ weight of i th attachment
 $y_i \Rightarrow$ deflection at the i th body location

(Rayleigh's equation)

the use of influence coefficients

$$\delta_{ij} = \begin{cases} \frac{b_j \cdot x_i}{6EIL} (l^2 - b_j^2 - x_i^2) & x_i \leq a_j \\ \frac{a_j(l-x_i)}{6EIL} (2lx_i - a_j^2 - x_i^2) & x_i > a_j \end{cases}$$



$\delta_{ij} = \delta_{ji}$

an influence coefficient is the transverse deflection at location i on a shaft due to a unit load at location j on the shaft

$$\begin{aligned} y_1 &= F_1 \delta_{11} + F_2 \delta_{12} + F_3 \delta_{13} \\ y_2 &= F_1 \delta_{21} + F_2 \delta_{22} + F_3 \delta_{23} \\ y_3 &= F_1 \delta_{31} + F_2 \delta_{32} + F_3 \delta_{33} \end{aligned}$$

F_i is because of w_i or $m_i \omega^2 y_i$

$$\begin{vmatrix} m_1 \delta_{11} - 1/\omega^2 & m_2 \delta_{12} & m_3 \delta_{13} \\ m_1 \delta_{21} & (m_2 \delta_{22} - 1/\omega^2) & m_3 \delta_{23} \\ m_1 \delta_{31} & m_2 \delta_{32} & (m_3 \delta_{33} - 1/\omega^2) \end{vmatrix} = 0$$

$\frac{1}{\omega_1^2} = \sum_{i=1}^n \frac{1}{\omega_{ii}^2}$ (neglect $1/\omega_2^2, 1/\omega_3^2, \dots$)
 Dunkerley's approximation

Consider three loads $P, Q,$ and R at positions 1, 2, and 3 on the shaft,

$$\omega_{11}^2 = \frac{g}{P \cdot \delta_{11}} \quad \omega_{22}^2 = \frac{g}{Q \cdot \delta_{11}} \quad \omega_{33}^2 = \frac{g}{R \cdot \delta_{11}}$$

$$\frac{1}{\omega_1^2} = \frac{1}{\omega_{11}^2} + \frac{1}{\omega_{22}^2} + \frac{1}{\omega_{33}^2} = \frac{P+Q+R}{g} \cdot \delta_{11}$$